

Revisiting acoustical gas-mixture separation

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1 This study experimentally investigated acoustically driven gas-mixture separation.
2 Acoustic wave propagation in a narrow tube can induce gas-mixture separation. A
3 binary mixture of helium and argon was used as the gas mixture. The pressure
4 amplitude of the acoustic wave and initial molar fraction of the helium gas were in-
5 vestigated. The obtained experimental data indicated that the molar fraction initially
6 increased with increasing pressure amplitude, whereas the saturated molar fraction
7 did not show a clear dependence on the pressure. Although the degree of separation
8 was smaller with purer helium, gas-mixture separation occurred under all conditions
9 within the experimental range.

10 I. INTRODUCTION

11 The propagation of an acoustic wave in a narrow tube results in the formation of a tem-
12 perature gradient in the radial direction owing to thermal interactions based on the following
13 principle: The compression/expansion of a gas induced by acoustic wave propagation in a
14 tube rises/lowers the temperature of the gas at the center of the tube. By contrast, owing
15 to the relatively large heat capacity of the tube wall, the gas temperature near the tube wall
16 is anchored to that of the tube wall. This acoustically generated temperature gradient can
17 contribute to energy conversion ([Biwa *et al.*, 2004](#)), mass transfer ([Weltsch *et al.*, 2017](#)), and
18 gas-mixture separation ([Spoor and Swift, 2000](#)).

19 This study focuses on the acoustical gas-mixture separation that was investigated in detail
20 by a research group at Los Alamos National Laboratory(LANL). The group briefly explained
21 the mechanism of acoustic gas separation and derived a theory ([Swift and Spoor, 1999](#)),
22 ([Geller and Swift, 2002a](#)), ([Geller and Swift, 2009](#)). This theory indicates that the separation
23 depends on the characteristics of the acoustic wave, such as the amplitude of the pressure or
24 velocity oscillations and the phase difference between them. It also depends on the geometry
25 of the tube, such as the tube radius. The LANL group conducted experiments using a
26 helium-argon gas mixture and quantitatively confirmed these results. However, few reports
27 except those from the LANL group are available. Hence, in this study, we revisit acoustical
28 gas-mixture separation experimentally and focus on two important parameters. The first
29 parameter is the amplitude of the acoustic waves. In theory, increasing the amplitude
30 is essential for separation. Therefore, we used a tube with a length comparable to the

31 wavelength of a sound wave to utilize acoustic resonance. This enabled an increase in the
 32 pressure amplitude to approximately 9 kPa in this study, which was 9% of the time-averaged
 33 pressure of the gas mixture charged in our experimental setup and was 4.5 times larger than
 34 that of the experiments by the LANL group (Spoor and Swift, 2000). The second parameter
 35 is the initial molar fraction of the helium and argon gases;

$$n_{He} = \frac{N_{He}}{N_{Ar} + N_{He}}, \quad (1)$$

36 where N_{He} and N_{Ar} are the mole numbers of the helium and argon gases in the experimental
 37 setup, respectively. Although increasing the separation rate in one separation step is essen-
 38 tial, multi-stage separation is also considered as an effective method for obtaining higher
 39 helium concentrations. Therefore, the dependence of the effect of acoustical gas-mixture
 40 separation on n_{He} is important.

41 The paper is organized as follows. Section II describes the experimental setup, proce-
 42 dures, and measurement methods used to determine the molar fraction of the gas mixture.
 43 Experimental results focusing on the time dependence of the molar fraction, effects of the
 44 pressure amplitude, and initial molar fraction are shown and discussed in Section III. Finally,
 45 the conclusions are presented in Section IV.

46 II. EXPERIMENTAL SETUP AND PROCEDURE

47 A. Setup

48 The experimental apparatus comprised a speaker unit and three types of tubes, as shown
 49 in Fig. 1. The speaker unit incorporated a nominal 6.5"/160 mm diameter moving-coil

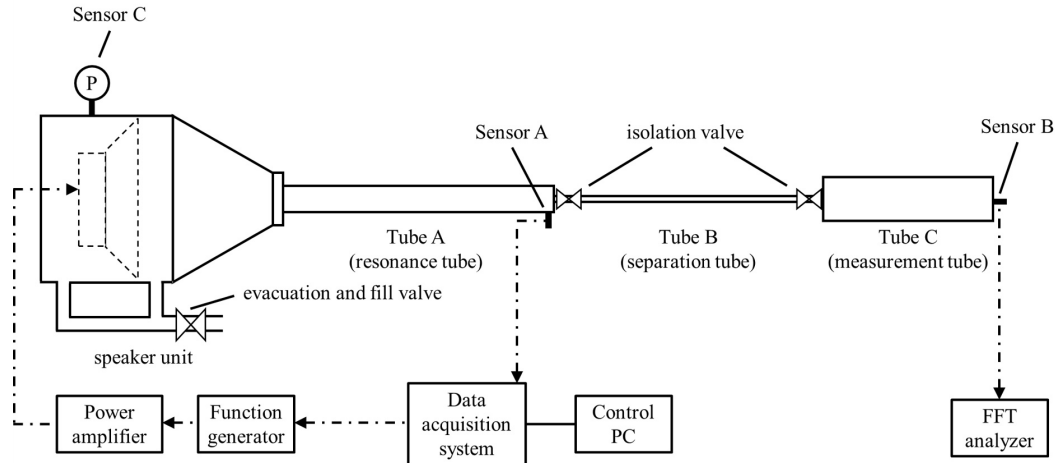


FIG. 1. Schematic of the experimental apparatus. Note that the dimensional ratio of each part (for example, the diameter/length ratio) differs from that of the actual apparatus.

50 electrodynamic loudspeaker with an effective piston area of 133 cm^2 (FW168HS, Fostex
 51 Ltd.) and was connected to Tube A by a tapered part. The volume between the diaphragm
 52 and Tube A was $1.09 \times 10^6 \text{ mm}^3$, and the volume behind the diaphragm was 1.59×10^6
 53 mm^3 . The input sinusoidal waveform was generated using a function generator (Agilent
 54 Technologies 33210A) and amplified using a 350 W stereo audio power amplifier (P2500S,
 55 Yamaha Corporation). The inner radius and length of Tube A were 11.5 mm and 2.0 m,
 56 respectively. The other end of Tube A was closed using a rigid plate with a small hole. A
 57 copper tube, referred to as Tube B, was connected to the hole and had the same diameter
 58 as the outer diameter of Tube B. Because the inner radius of the tube was the narrowest
 59 in the apparatus, the gas-mixture separation was expected to occur mainly in this part.
 60 The inner radius and length of Tube B were 2.4 mm and 1.86 m, respectively. Tube C was
 61 connected to the other end of Tube B. The inner radius and length of Tube C were 18.9

62 mm and 0.148 m, respectively. The apparatus had two types of valves: one (evacuation
63 and fill valve) was used to separate the gas inside and outside, and the other was used
64 to isolate Tube C. Two types of sensors were mounted on the apparatus. Sensors A and
65 B (PD104, Jtect Ltd.) were used to measure the acoustic pressure amplitude at the ends
66 of Tubes A and C, respectively. These sensors were piezoresistive, and one side of the
67 diffusion-type gauge was opened to the atmosphere. Both steady and acoustic pressures
68 were measured, and the amplitude corresponding to the driving frequency was extracted.
69 Sensor C (GP-M025, Keyence Corporation) was mounted on the speaker unit to monitor
70 the internal mean pressure. The signals from Sensors B and C were analyzed using a fast
71 Fourier transform analyzer (DS-3000, Onosokki Ltd.). As described in the next section,
72 the sinusoidal input waveform to the speaker unit was adjusted according to the progress
73 of separation. Therefore, the function generator and Sensor A were connected to a data
74 acquisition system (USB-6363 and BNC-2120, National Instruments Corporation), and the
75 amplitude and frequency of the input waveform were feedback-controlled.

76 The working gas was a binary mixture of helium and argon. The temperature of the
77 experimental apparatus was controlled to room temperature (approximately 20°C) using a
78 normal air conditioner. The room temperature fluctuated by approximately ± 1 °C dur-
79 ing the experiment. Note that throughout the experiment, the local temperature in the
80 apparatus was subject to change owing to the thermoacoustic effect.

B. Procedure

Five preliminary steps were performed before the experiments. First, the gas in the experimental setup was vacuumed from the evacuation-and-fill valve under the condition that all valves were open, and then the two isolation valves were closed. Second, helium and argon gases were injected sequentially into the apparatus. The amount of injected gas was monitored using Sensor C. The time-averaged pressure inside the apparatus p_m was approximately 100 kPa. The two gases did not mix immediately after injection and were unevenly distributed in the tube. Third, an acoustic wave was input from the speaker to forcibly mix the two gases in Tube A. The progress of the binary gas mixing was checked using the first resonance frequency in Tube A. Preliminary experiments showed that the resonance frequency in Tube A corresponded to the molecular weight of the initially injected gas and gradually approached the frequency corresponding to the average molecular weight of the mixed gas. In this procedure, the first-resonance-frequency sound wave with a pressure amplitude of 3.0 kPa was applied. The driving (resonance) frequency was adjusted and observed once every 3 min during the forcible mixing. When the frequency change became less than 0.25, the sound wave input was stopped, and the resonant frequency continued to be measured once every 3 min. Figure 2 shows the time dependence of the resonance frequency in Tube A. The mixing procedure required approximately 1 h. After mixing, the experimental apparatus was allowed to stand for approximately one day. Fourth, the two isolation valves were opened, and the mixed gas was introduced into Tubes B and C. Fifth, the molar fraction in Tube C, $n_{C,He}$, was measured according to the method described in

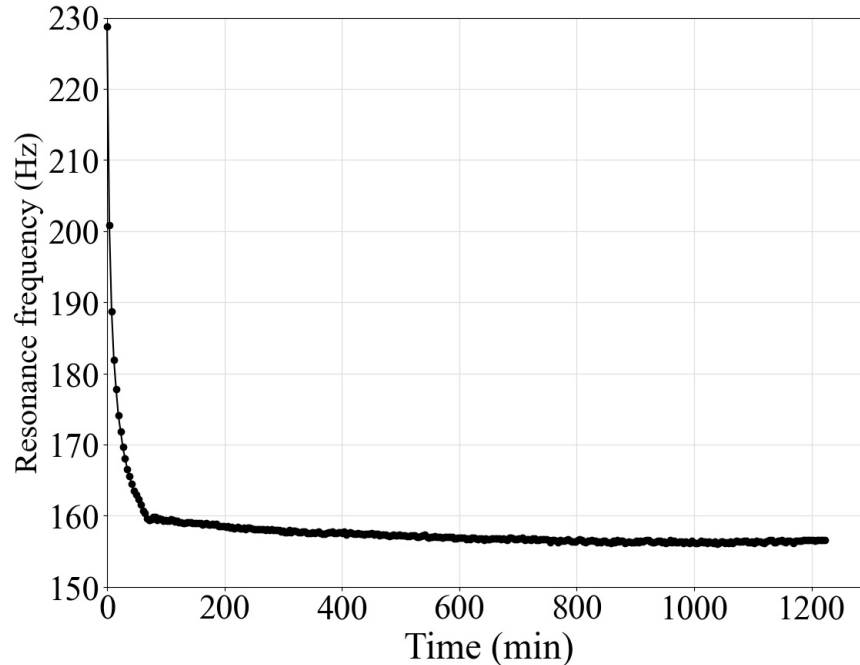


FIG. 2. Time dependence of the resonance frequency in Tube A. The depicted case used the gas mixture with a helium initial molar fraction of 50.0 %.

103 the next subsection and compared with the value estimated from the amounts of the two
 104 gases injected in the second step. If the two values were consistent, the measured $n_{C,He}$ was
 105 considered as the initial molar fraction $n_{0,He}$.

106 After the preparations, an acoustic wave with the second resonance frequency f was
 107 applied under the condition that both isolation valves were opened, and gas separation was
 108 initiated. Although we do not discuss the effect of the mode of the resonance frequency
 109 on the gas-separation capability in this study, preliminary experiments showed that an
 110 acoustic wave with the second resonance frequency had a better gas-separation capability
 111 than that with the first resonance frequency. Therefore, we focused on the second resonance.
 112 Because the resonance frequency f gradually changed as the separation progressed, the

113 driving frequency of the speaker was adjusted. Here, the resonance frequency is defined as
 114 the frequency that generates the maximum pressure amplitude at Sensor A with the same
 115 input voltage amplitude to the loudspeaker. The resonance frequency was determined by
 116 sweeping the driving frequency. The input voltage amplitude to the speaker was adjusted
 117 using the determined resonance frequency such that the pressure amplitude p_A at Sensor A
 118 was at the target value. The frequency and voltage amplitude were adjusted every 10 min.
 119 Note that the two valves were always open except during the measurement of the molar
 120 fraction $n_{C,He}$ in Tube C described in the next subsection.

121 C. Measurement method of molar fraction

122 The first resonance frequency f_{Ctube} in the cylindrical tube corresponds to $a_0/2l$, where a_0
 123 and l are the adiabatic sound speed and cavity length, respectively, when the valve between
 124 Tube B and Tube C is closed. Using the averaged molar weight m_{avg} of the gas, a_0 is
 125 expressed as $\gamma RT_m/m_{avg}$, where γ is the specific heat ratio, R is the universal gas constant,
 126 and T_m is the time-averaged absolute temperature. Using these two equations, the following
 127 equation is derived:

$$m_{avg} = \gamma RT_m \left(\frac{1}{2lf_{Ctube}} \right)^2 \quad (2)$$

128 In this study, m_{avg} was expressed in another form as follows:

$$m_{avg} = m_{He}n_{He} + m_{Ar}(1 - n_{He}), \quad (3)$$

129 where m_{He} and m_{Ar} are the molar weights of helium and argon, respectively. Transforming
 130 these equations yields the following equation for the helium molar fraction:

$$n_{He} = \frac{m_{Ar} - m_{avg}}{m_{Ar} - m_{He}} \quad (4)$$

131 Note that the thermal boundary layer thickness calculated using equation (A4) in Appendix
 132 has a maximum value of 0.77% at approximately $n_{He} = 0.8$. This value indicates that the
 133 boundary layer correction can be neglected in Tube C.

134 The measurement procedure for n_{He} is as follows: First, both isolation valves were closed.
 135 Second, the gas column inside Tube C was excited using an impact hammer, and a sensor
 136 signal was obtained. Third, the signal was analyzed using the fast Fourier transform ana-
 137 lyzer, and the first resonance frequency of the gas column in Tube C was obtained. Finally,
 138 the molar fraction in Tube C $n_{C,He}$ was estimated by substituting the obtained resonance
 139 frequency into Eqs. (2) and (4). The error in measuring the molecular weight using this
 140 method was confirmed to be approximately 0.58 for air, 0.67 for pure helium, and 0.33
 141 for pure argon. This result indicates that the current method can measure the molecular
 142 weight with an error of approximately 0.7 regardless of the separation state. This error
 143 in the molecular weight corresponds to an approximately 1.9% molar fraction error for the
 144 He-Ar gas mixture.

145 III. RESULTS AND DISCUSSIONS

146 A. Time dependence of molar fraction

147 In this subsection, one of the experimental results we obtained is taken as an example to
 148 show the time dependence of gas separation. The initial molar fraction $n_{0,He}$ was 0.50, and
 149 the pressure amplitude p_A was set at 3.0 kPa. The driving frequency corresponding to the
 150 second mode was approximately 159 Hz at the beginning of the experiment. The values of
 151 p_A and $n_{0,He}$ are similar to those in the experimental study by the LANL group (Spoor and
 152 Swift, 2000), although the frequency in this study is much higher than that in their study.
 153 The sound wave with this frequency in the current-mixed gas had a wavelength comparable
 154 to that of Tube B.

155 Figure 3 displays the time dependence of $n_{C,He}$. First, the data near $t = 0$ is considered.
 156 Figure 3 shows that $n_{C,He}$ near $t = 0$ rapidly increases. Here, the experimental rate of
 157 increase $\dot{n}_{C,He}$ is defined as

$$\dot{n}_{C,He}(t_i) = \frac{n_{C,He,i+1} - n_{C,He,i}}{t_{i+1} - t_i}, \quad (5)$$

158 where i is the index of the acquired data. The molar flow into Tube C can be calculated
 159 using $\dot{n}_{C,He}$ and the volume of Tube C, V_C , as

$$\dot{N}_{He}(t_i) = \dot{n}_{C,He}(t_i) \times \frac{p_m V_C}{RT_m(t_i)}, \quad (6)$$

160 Processing data shown in Fig. 3, $\dot{N}_{He}(0)$ of this experiment were determined to be 3.0×10^{-8}
 161 mol/s, which is comparable to the value reported by the LANL group (Spoor and Swift,
 162 2000).

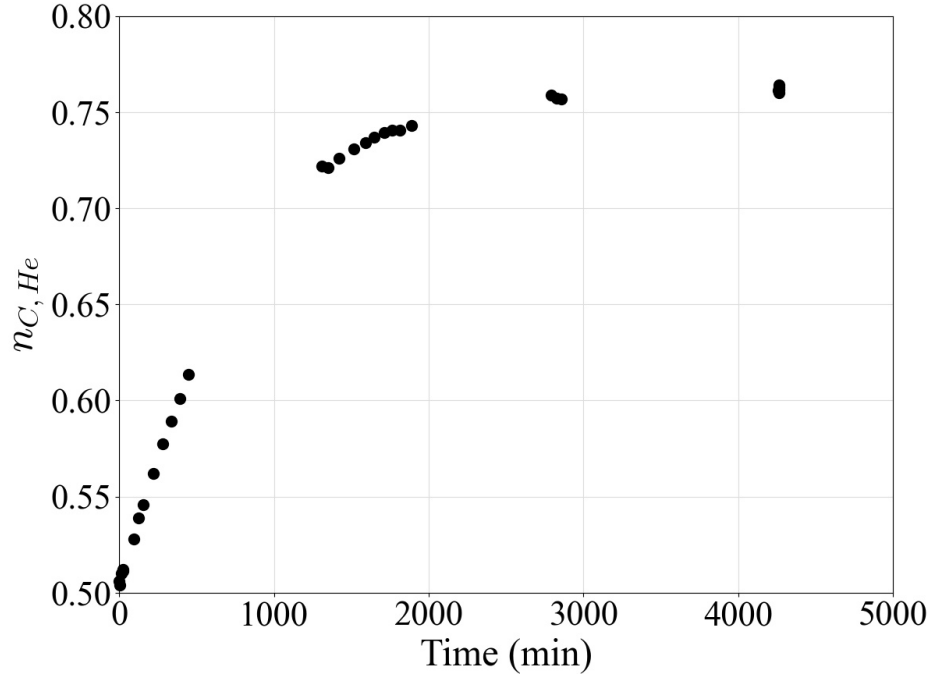


FIG. 3. Time dependence of the mole fraction $n_{C, He}$. The gas-separation experiment was conducted under the condition that the initial molar fraction $n_{0, He}$ was 0.50, and the pressure amplitude p_A was 3.0 kPa.

163 Next, we focused on the data under saturated conditions. Figure 3 shows that the
 164 saturated $n_{C, He}$ is 76.0%. Now, there are two common boundary conditions throughout the
 165 experiment: the velocity at the end of Tube C was zero, and the pressure amplitude at the
 166 beginning of Tube B was set to the target value (3.0 kPa in this case). In the saturated
 167 condition, the driving frequency was 159.2 Hz, and the molar fraction at the end of Tube C
 168 was 76.0 % under the assumption that the molar fraction in Tube C was constant. Under
 169 these conditions, the acoustic and molar-fraction distributions in the saturated condition
 170 can be estimated as shown in Fig. 4. The commonly used thermoacoustic software DeltaEC
 171 was used for this calculation (Ward *et al.*).

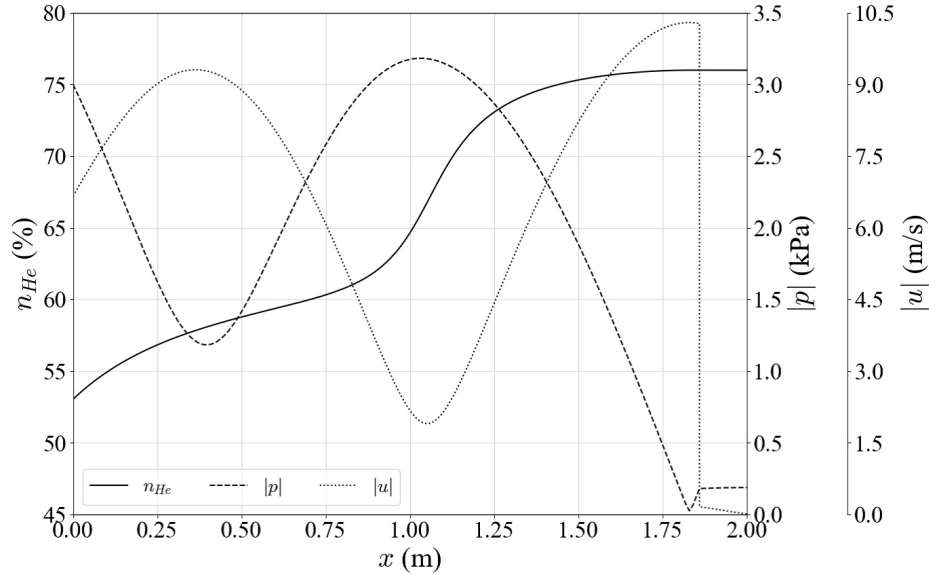


FIG. 4. Estimated acoustic and molar-fraction distributions in Tube B and C at the saturated condition. The commonly used thermoacoustic software DeltaEC was used for this calculation (Ward *et al.*).

172 Figure 4 shows a notable molar-fraction gradient in Tube B, while the molar fraction
 173 is almost constant in Tube C (as expected). The molar fraction in Tube B does not vary
 174 linearly in the axial direction, and the gradient is locally large near the antinode of the
 175 pressure amplitude (node of the velocity amplitude). The estimated molar fractions at the
 176 beginning and end of Tube B are 53.0 % and 76.0 %, respectively. Therefore, the gradient
 177 of n_{He} along Tube B is 12.4(% /m), which is also comparable to the data of the LANL
 178 group (Spoor and Swift, 2000). Therefore, a tube with a length comparable to the acoustic
 179 wavelength can contribute to acoustic gas-mixture separation. These results motivate us to
 180 perform acoustical gas-mixture separation using a thermoacoustic engine that has no moving
 181 parts and can be powered by an external heat source.

182 The increased number of molecules in Tube C originates from the space to the left of
183 Tube B, which includes Tube A. Assuming that the molar fraction is constant in the spaces
184 to the left of Tube B and also in Tube C, the decrease in the molar concentration in Tube
185 A is estimated to be approximately 1.2% based on the volume ratio and the increase in the
186 molar fraction by 26 % in Tube C. Surprisingly, however, the calculation result displayed
187 in Fig. 4 shows that the molar fraction at the beginning of Tube B (i.e., the end of Tube
188 A) increases from the initial value of 50 %. This suggests that a molar-fraction distribution
189 occurs in Tube A, even though the tube radius is larger than Tube B.

190 If appropriate boundary conditions in the final state are available, the acoustic and molar-
191 fraction fields in the final state can be determined in advance. However, to the best of the
192 author's knowledge, it is impossible to determine the appropriate boundary conditions for
193 the final state in advance, which implies that the final separation state cannot be predicted.
194 This is caused by the change in the molar-fraction distribution during the gas-mixture
195 separation. Overcoming this issue requires the knowledge of the molar fraction distribution
196 after the start of separation; namely, it requires a time-evolution solver.

197 **B. Effect of pressure amplitude**

198 Because the acoustic wave was driven by the resonance frequency in this experiment, p_A
199 can easily be set to a large value. Acoustical gas-mixture separation was performed with
200 $p_A = 1.5, 3, 6,$ and 9 kPa. Here, we focus on two quantities: \dot{N}_{He} near $t = 0$ and Δn_{He} .

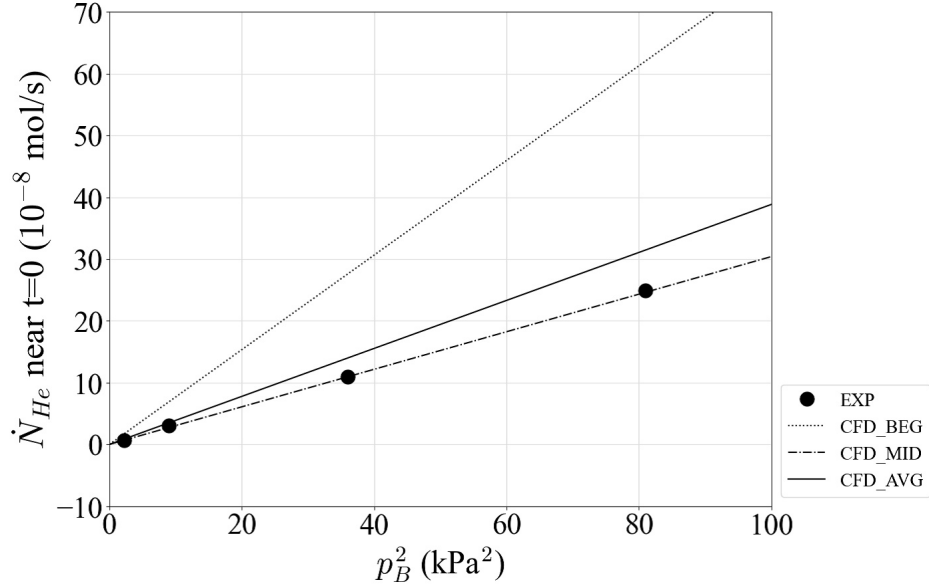


FIG. 5. Pressure-amplitude dependence of \dot{N}_{He} near $t = 0$. Note that the values are plotted as a function of p_B^2 . Black-filled circles display the values estimated from the experimental results by Eq. (5) and (6). The calculated \dot{N}_{He} at the beginning (CFD_BEG), midpoint (CFD_MID), and the overall average (CFD_AVG) of Tube B at $t = 0$ are plotted as a function of the pressure amplitude with lines.

201 Figure 5 shows \dot{N}_{He} near $t = 0$ as a function of p_A^2 . The black circles display the experi-
 202 mental results, which indicate that \dot{N}_{He} increases linearly with p_A^2 within the experimental
 203 range.

204 Geller and Swift derived the theory and equation for \dot{N}_{He} in (Geller and Swift, 2002a)
 205 and arranged it as Eq. (A1) in (Geller and Swift, 2004). In the initial state (before gas
 206 separation was commenced), there was no molar fraction gradient in the apparatus, and
 207 dn_{He}/dx could be considered as zero everywhere. In addition, because this experiment was
 208 conducted in a sealed system, the total molar flux was zero, that is, $\dot{N}_{He} = -\dot{N}_{Ar}$ at all

209 locations, where \dot{N}_{Ar} indicates the molar flux of argon. Therefore, the equation can be
 210 expressed as:

$$\frac{\dot{N}_{0,He}}{A_{gas}} = -\frac{\delta_\alpha}{4r_h} \frac{\gamma - 1}{\gamma} \frac{k_T}{R_{univ}T_m} |p||u|[F_{trav} \cos \theta + F_{stand} \sin \theta] \quad (7)$$

211 where A_{gas} denotes the cross-sectional area of a flow channel, δ_α denotes the thermal bound-
 212 ary layer thickness, r_h denotes the tube radius, γ denotes the specific heat ratio, k_T denotes
 213 the thermal diffusion ratio which is proportional to the driving force of the Soret effect
 214 (reviewed in detail in (Platten, 2006) and (Rahman and Saghir, 2014)), R_{univ} denotes the
 215 universal gas constant, and θ denotes the phase difference at which p leads u . The definitions
 216 of F_{trav} , F_{stand} , and related values are provided in the Appendix. All the values except p , u ,
 217 and θ depend only on the current molar fraction. Therefore, if the distributions of p and
 218 u are obtained, the $\dot{N}_{0,He}$ distribution in the initial state can be calculated. We estimated
 219 the p and u distributions based on the linear acoustic theory (Swift, 2003), including the
 220 evolution of u (Geller and Swift, 2002b) modified for gas-separation calculations and the
 221 effect of a minor loss (Ueda *et al.*, 2020) occurring at the junction of Tubes B and C. Figure
 222 6 shows the calculated distributions of \dot{N}_{He} at $t = 0$ as a function of the axial position. The
 223 plotted cases correspond to the experimental conditions. The magnitude is the largest near
 224 the beginning of the tube and decreases toward the end although there is a local flat dis-
 225 tribution. The basic distribution shape remains almost the same regardless of the pressure
 226 amplitude in the calculations according to the linear thermoacoustic theory.

228 The numerically calculated \dot{N}_{He} of the beginning (CFD_BEG), midpoint (CFD_MID, x
 229 = 0.93 m), and overall average (CFD_AVG) of Tube B at $t = 0$ are plotted as a function
 230 of the pressure amplitude with lines in Fig. 5. The experimental and numerical results

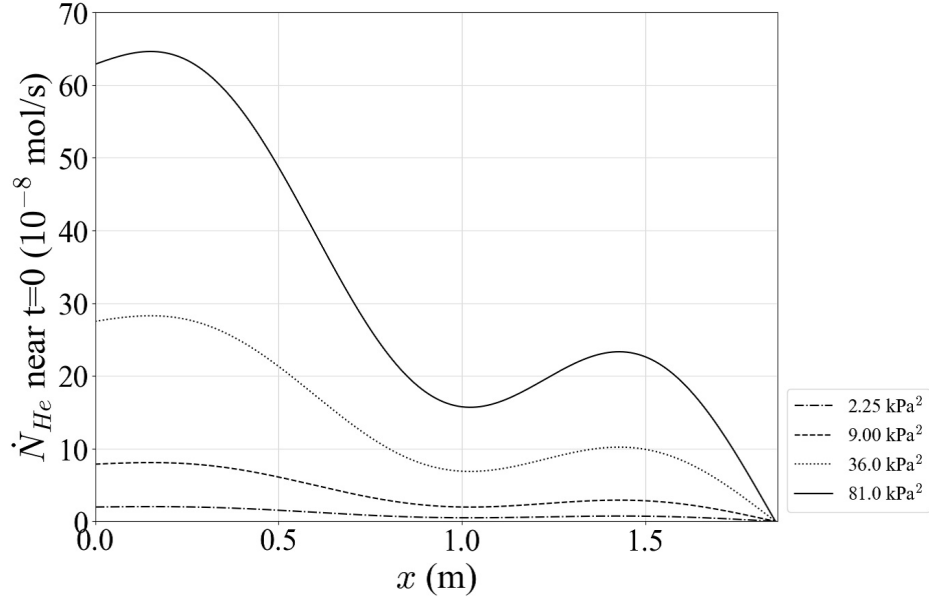


FIG. 6. Distributions of \dot{N}_{He} at $t = 0$. The plotted cases correspond to the experimental conditions in Fig. 5.

231 qualitatively show the same trend: \dot{N}_{He} increases linearly with p_b^2 . Although the result of
 232 the midpoint of Tube B matches well with that of the experiments, we consider this to be
 233 coincidental because the \dot{N}_{He} distribution can change depending on the tube length and
 234 separation progress. In addition, the experimental results represent the average value of
 235 the change over 10 min from the start, whereas the calculations represent the local value at
 236 the initial state. The following two points are important in this plot. The first is that the
 237 experimentally obtained results are similar to the order of those obtained by calculations.
 238 The second is that \dot{N}_{He} changes linearly with respect to the pressure amplitude in the
 239 experiment and calculation.

240 Figure 7 displays Δn_{He} as a function of p_B^2 . This figure shows that Δn_{He} increases with
 241 increasing p_B^2 when p_B^2 is less than or equal to $(3 \text{ kPa})^2$. In contrast, when p_B^2 is greater than

242 or equal to $(6 \text{ kPa})^2$, Δn_{He} decreases. Note that the theoretical Δn_{He} is difficult to obtain
 243 because the position-dependent n_{He} in the setup changes as the gas separation progresses
 244 and is not constant along Tube B. Based on the experimental results, an excessively large
 245 pressure amplitude is not suitable for acoustical gas-mixture separation, and the appropriate
 246 pressure amplitude to obtain a larger Δn_{He} is 6% of p_m (6 kPa) in this experiment. The
 247 LANL group observed a decrease in Δn_{He} and concluded that this was due to mixing by
 248 acoustic streaming (Geller and Swift, 2002a). Another possibility is acoustic turbulence. As
 249 described in the introduction, the generation of a temperature gradient in the radial direction
 250 due to the compression/expansion of gas and the movement of gas molecules due to the Soret
 251 effect are very important in gas-mixture separation. In the range of the conventional linear
 252 thermoacoustic theory, the flow in the tube is assumed to be laminar, and there is no flow
 253 in the radial direction. Under these conditions, the Soret effect is the only driving force for
 254 gas-molecule movement in the radial direction. However, acoustic waves with a very large
 255 amplitude cause acoustic turbulence and the "laminar flow" assumption cannot be applied.
 256 Turbulent flow causes a radial-direction flow, which may cause gas-particle movement due
 257 to convection and lead to the cancellation of the biased gas-molecule distribution due to
 258 the Soret effect. Ohmi et al. summarized the regimes of oscillating flow by the square root
 259 of the dimensionless frequency $\omega' = r^2\omega/\nu$ and acoustic Reynolds number Re_{os} (Ohmi and
 260 Iguchi, 1982), where ω denotes the angular frequency of an acoustic wave, and ν denotes
 261 the kinematic viscosity. The acoustic Reynolds number is defined as follows:

$$Re_{os} = \frac{2r|u|}{\nu} \quad (8)$$

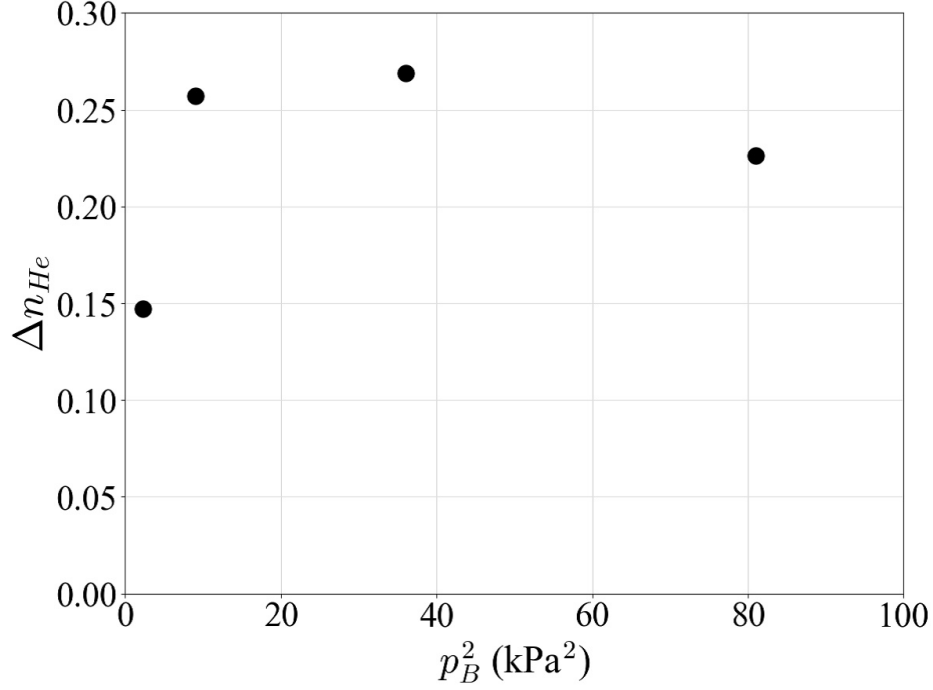


FIG. 7. Pressure-amplitude dependence of Δn_{He} as a function of p_B^2

262 Under the condition that the pressure amplitude is 9 kPa in Fig. 7, ω' and Re_{os} are 15.2
 263 and 4.22×10^3 , respectively. This condition falls in the transient region between the laminar
 264 and turbulent regions. Therefore, there is a possibility that the effect of acoustic turbulence
 265 has appeared to some extent, leading to a decrease in the gas-mixture separation under this
 266 condition.

267 C. Effect of initial molar fraction

268 As demonstrated in Section III B, there is a limitation to how much $n_{C,He}$ can be in-
 269 creased by increasing p_B . Hence, to obtain high-purity helium gas from a He-Ar mixture,
 270 we assume a different method: multi-stage gas-mixture separation. Considering a two-stage

271 gas separation as an example, the first-stage process generates a middle n_{He} mixture from a
 272 low one, and then the second-stage process generates a higher n_{He} mixture from the middle
 273 one. Because this method considers gas-mixture separation at multiple mole fractions, the
 274 effects of $n_{0,He}$ on \dot{N}_{He} and Δn_{He} are significant. Hence, the initial molar fraction $n_{0,He}$
 275 were set to 0.28, 0.70, and 0.90, respectively, and $\dot{n}_{C,He}$ near $t = 0$ and Δn_{He} were measured
 276 under saturated conditions. During the experiments, p_B was maintained at 6 kPa, and the
 277 driving frequency was adjusted to the second resonance frequency. The frequency changed
 278 according to the value of $n_{0,He}$.

279 Figure 8 shows \dot{N}_{He} near $t = 0$ as a function of $n_{0,He}$. The dashed line is calculated in the
 280 same manner as shown in Fig. 5. The experimental data at $n_{0,He} = 0.50$ are the same as
 281 those presented in Section III B. The figure shows that the experimental values of \dot{N}_{He} are of
 282 similar magnitudes, indicating that the acoustic gas-mixture separation can work in a wide
 283 range of $n_{0,He}$. Moreover, the experimental and computational results exhibit a maximum
 284 at approximately $n_{0,He} = 0.70$.

285 We consider the dependence of the thermal diffusion ratio k_T on n_{He} in Eq. (7) as the
 286 main reason. Atkins et al. experimentally investigated k_T and proposed an approximate
 287 formula (Atkins *et al.*, 1939). Figure 9 shows a plot of k_T as a function of n_{He} along a solid
 288 line, where k_T is calculated using the approximate formula of the *Helium – argon* section
 289 of the above reference. This figure indicates that k_T is zero at $n_{He} = 0.0$ and 1.0 and has
 290 a clear peak at approximately $n_{He} = 0.65$. In contrast, although the products of the other
 291 values, except for k_T in Eq. (7), depend on n_{He} , the difference is approximately twice as

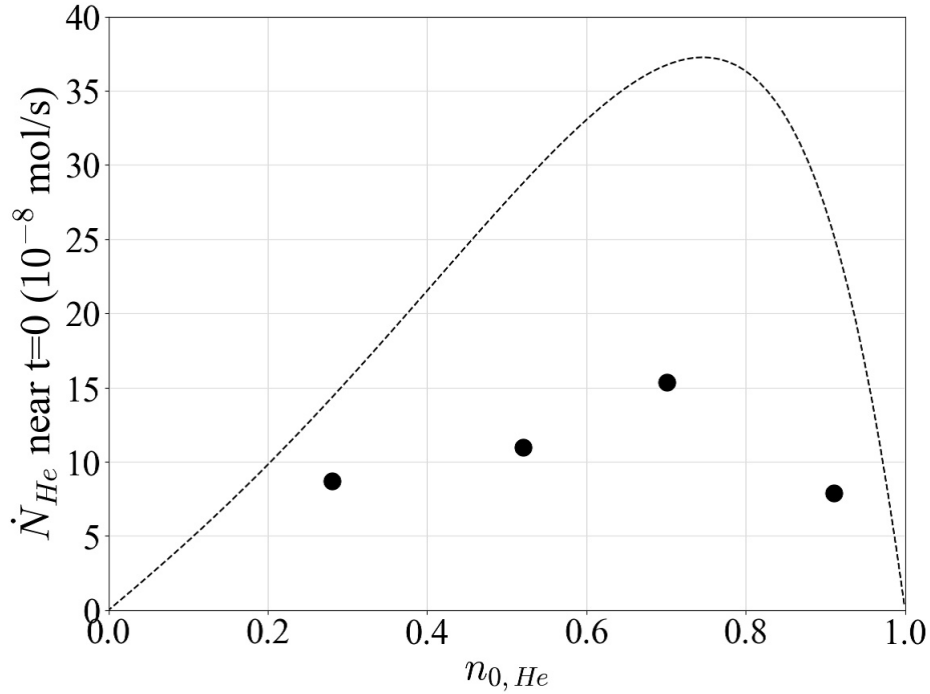


FIG. 8. Initial-molar-fraction $n_{0,He}$ dependence of the \dot{N}_{He} near $t = 0$. Black-filled circles and a dashed line are calculated in the same manner as shown in Fig. 5.

292 large. Therefore, it can be concluded that k_T mainly depends on the initial molar fraction,
 293 as shown in Fig. 8.

294 Figure 10 shows Δn_{He} as a function of $n_{0,He}$. The black circles display the experimental
 295 results, and the solid line is the upper limit of Δn_{He} (e.g. when $n_{0,He}$ is 0.7, the upper
 296 limit of Δn_{He} is 0.3). This figure shows that Δn_{He} is 0.2 when $n_{0,He}$ is 0.70, implying that
 297 $n_{C,He}$ under the saturated condition is 0.90, and $n_{C,He} = 0.98$ can be obtained from the gas
 298 mixture of $n_{0,He} = 0.90$. Although the degree of separation was smaller with purer helium,
 299 gas-mixture separation occurred under all conditions within the experimental range. The

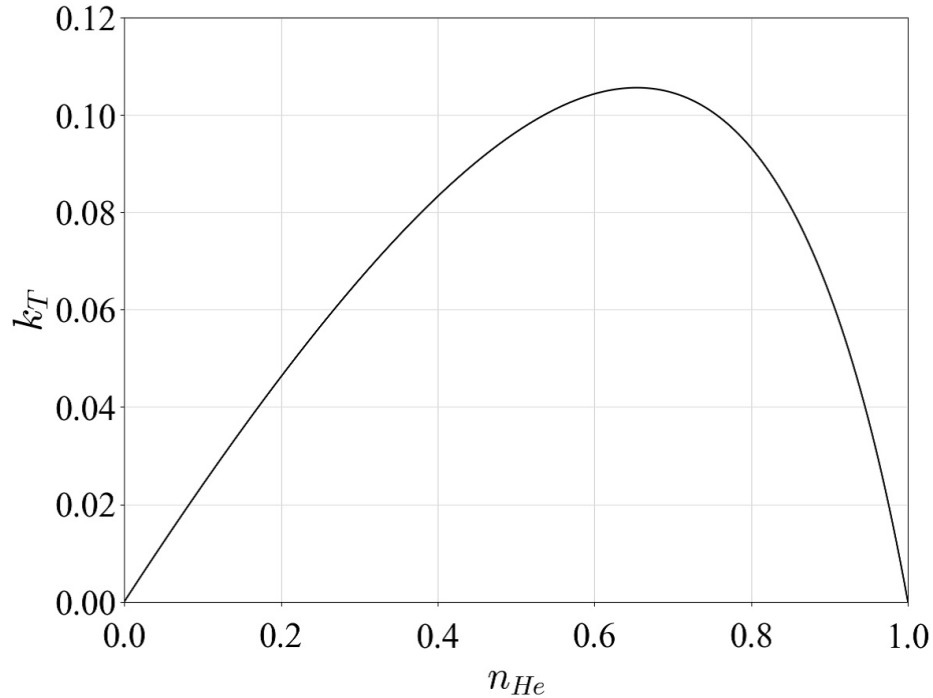


FIG. 9. Molar-fraction n_{He} dependence of the thermal diffusion ratio k_T from (Atkins *et al.*, 1939)

300 results indicate that a multi-stage gas-mixture separation system can be used to obtain purer
 301 gas.

302 IV. SUMMARY AND CONCLUSIONS

303 This study experimentally investigated acoustical gas-mixture (helium and argon) separa-
 304 tion. We focused on two important parameters for the gas-mixture separation: the pressure
 305 amplitude of the acoustic wave and the initial molar fraction. The results showed that our
 306 apparatus could induce gas-mixture separation. A gradual increase in the molar fraction
 307 in the measurement tube was observed. Although the experimental conditions were not
 308 the same as those in a previous study by the LANL group, the initial molar flux and final

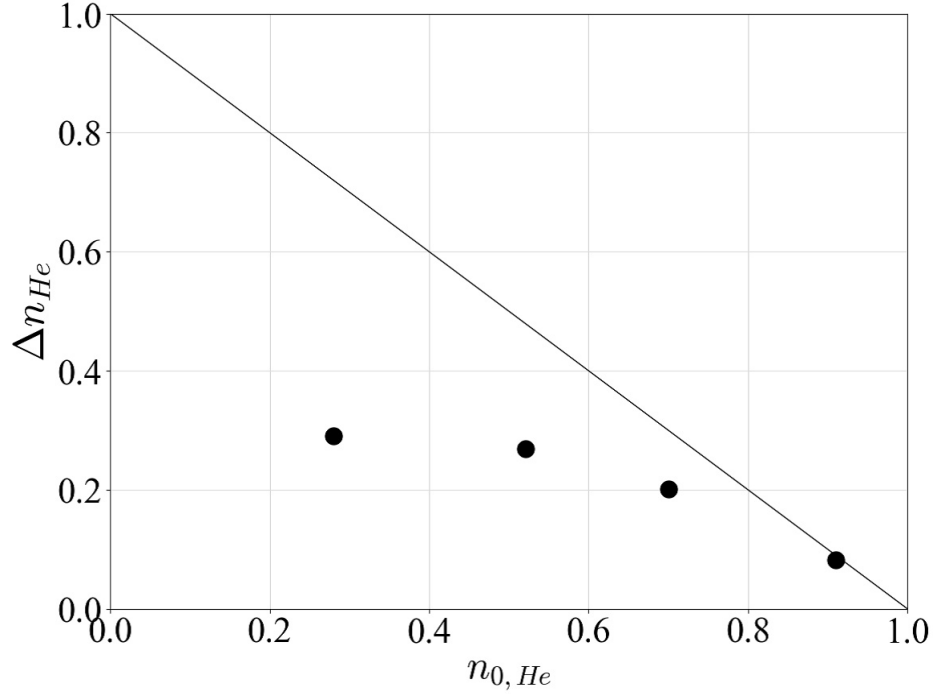


FIG. 10. Initial-molar-fraction $n_{0,He}$ dependence on Δn_{He} . A solid line shows the upper limit of Δn_{He} .

309 molar-fraction gradient of our experiment were confirmed to be comparable. Increasing the
 310 pressure amplitude increased the initial molar flux. However, the degree of final separation
 311 had a peak value, and a larger pressure amplitude did not always result in better separa-
 312 tion. The initial molar flux depended on the initial molar flux and peaks at an initial molar
 313 fraction of approximately 0.7. This suggests that the thermal diffusion ratio significantly
 314 affected the initial molar flux. Although the degree of separation was smaller with purer
 315 helium, gas-mixture separation occurred under all conditions within the experimental range.
 316 This result indicates that a multi-stage gas-mixture separation system can be used to obtain
 317 purer gas.

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322 **AUTHOR DECLARATIONS**323 **Conflict of Interest**

324 The authors declare no conflict of interest.

325 **Data Availability**

326 Data are available from the authors upon request.

327 **APPENDIX: CALCULATION OF F_{TRAV} AND F_{STAND}**

328 F_{trav} and F_{stand} are defined as follows (Geller and Swift, 2004):

$$F_{trav} = -\frac{2r_h}{\delta_\alpha} \operatorname{Re} \left[\frac{G}{1 - \tilde{\chi}_\nu} \right] \quad (\text{A1})$$

$$F_{stand} = \frac{2r_h}{\delta_\alpha} \operatorname{Im} \left[\frac{G}{1 - \tilde{\chi}_\nu} \right] \quad (\text{A2})$$

329 $\chi_j(j = \nu, \alpha D, D\alpha)$ is a complex function for function for a circular tube flow channel defined
 330 as follows:

$$\chi_j = \frac{2J_1(Y_j)}{Y_j J_0(Y_j)} \quad (\text{A3})$$

331 where

$$Y_j = \frac{(i-1)r}{\delta_j} \quad (\text{A4})$$

332 and J_α , r , and δ_j ($j = \nu, \alpha, \alpha D, D\alpha$) are the Bessel functions of the α -th kind, the radius of
 333 the tube, and the boundary layer thickness, respectively. Each boundary layer thickness is
 334 defined as follows:

$$\delta_\nu = \sqrt{\frac{2\nu}{\omega}} \quad (\text{A5})$$

$$\delta_\alpha = \sqrt{\frac{2\alpha}{\omega}} \quad (\text{A6})$$

$$\delta_{\alpha D}^2 = \frac{1}{2}\delta_\alpha^2[1 + (1 + \varepsilon)/L + \sqrt{[1 + (1 + \varepsilon)/L]^2 - 4/L}] \quad (\text{A7})$$

$$\delta_{D\alpha}^2 = \frac{1}{2}\delta_\alpha^2[1 + (1 + \varepsilon)/L - \sqrt{[1 + (1 + \varepsilon)/L]^2 - 4/L}] \quad (\text{A8})$$

335 where

$$\varepsilon = \frac{\gamma - 1}{\gamma} \frac{k_T^2}{n_H(1 - n_H)} \quad (\text{A9})$$

$$L = \frac{\alpha}{D_{12}} \quad (\text{A10})$$

336 and ν , α , k_T , n_H , and D_{12} are the kinematic viscosity, thermal diffusivity, thermal diffusion
 337 ratio, molar fraction of the heavier component, and mutual diffusion coefficient, respectively.

338 G and related values are defined as follows:

$$S = \left(\frac{\delta_\alpha^2}{\delta_{D\alpha}^2} - 1 \right) \chi_{D\alpha} - \left(\frac{\delta_\alpha^2}{\delta_{\alpha D}^2} - 1 \right) \chi_{\alpha D} \quad (\text{A11})$$

$$Q = \frac{\delta_{\alpha D}^2 - \delta_{D\alpha}^2}{\delta_\alpha^2} \quad (\text{A12})$$

$$M = (1 + \sigma)(1 + \sigma L) + \varepsilon\sigma \quad (\text{A13})$$

$$G = \frac{\sigma L Q}{M S} \chi_{\alpha D} \chi_{D\alpha} + \frac{\tilde{\chi}_\nu}{S} \left(\frac{\chi_{\alpha D}}{1 + \delta_\nu^2/\delta_{D\alpha}^2} - \frac{\chi_{D\alpha}}{1 + \delta_\nu^2/\delta_{\alpha D}^2} \right) \quad (\text{A14})$$

339 where σ denotes the Prandtl number.

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